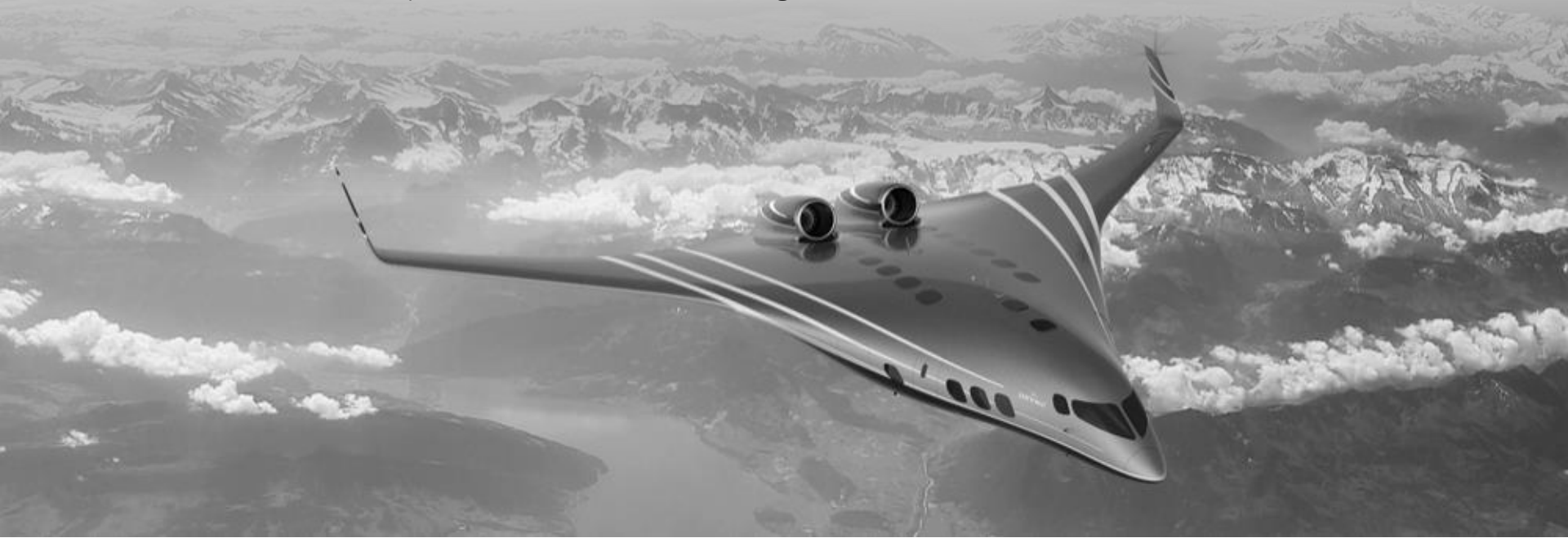


The Future of BWB

7th International Workshop on Aviation & Climate Change



Mark A. Page

Founder & Chief Scientist DZYNE Technologies



Blaine Rawdon
Mark Page
Bob Liebeck

Fathers of the Modern BWB (above)

Dr. Ben Tigner and Bob at right with
other Stanford Students and the
BWB-17 demonstrator



A Times Mirror Magazine

1995's Concept Cars

The New 800-Passenger Jetliners



*****5-DIGIT 90630
#30PGE 01980 3UN497# 5 JMA97
TENISE PAGE 00000 MM
9803 VIA SONOMA PPS 000006
CYPRESS CA 90630-3437
P04-1

A 2000 1 000

\$2.99

35662 0726597 04>

**WITH STEVEN UDVAR-HAZY** Page 57

YOU MADE IT START
HAPPEN BILL (LAW)

\$6.00 AUGUST 6, 2007

AVIATION WEEK

& SPACE TECHNOLOGY



MARK -
YOU ARE PART
OF THIS, THANKS
Bob

[illegible]

#ALH0001193705/0# R0603R001 23FEB09

00000

MARK A PAGE 009797

SWIFT ENGINEERING 00001

9897 ITe SoNOMe	1.777
-----------------	-------

CYPRESS Co. 99679-7477

The McGraw-Hill Companies

NEW ITIES UAVS

Page 48

MARK,
GOD BLESS YOU
AS WELL! MIKE KISLA

ESA'S EARTH EXPLORERS

Page 61

Mark,
You started this.
I Deneb En

www.aviationweek.com/awst

SEPTEMBER 26, 2016

AVIATIONWEEK

& SPACE TECHNOLOGY



X-Planes

BEYOND BOEING

AVIATION WEEK
NETWORK

Swift KillerBee and Northrop BAT



Ascent1000



Ascent1000





Ascent1000



Ascent1000



Super Regional Jet Interior



Ascent1000

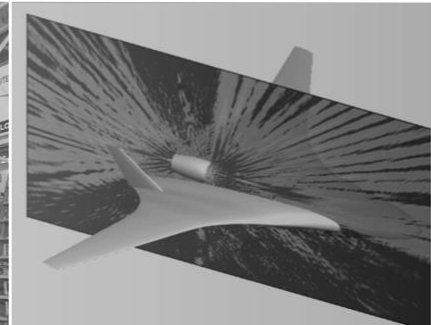
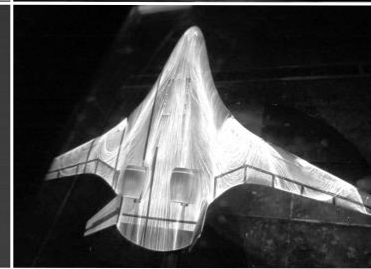
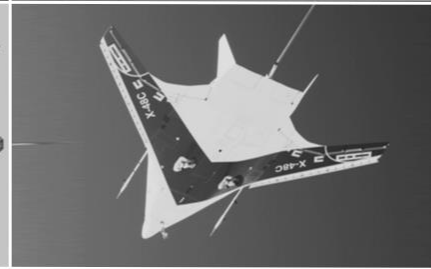
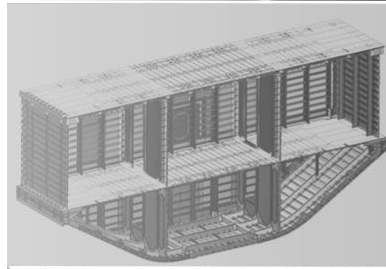
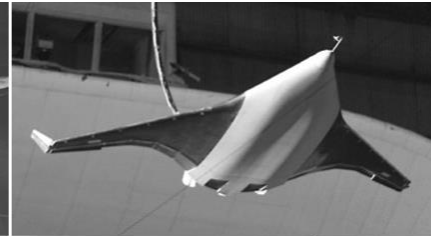
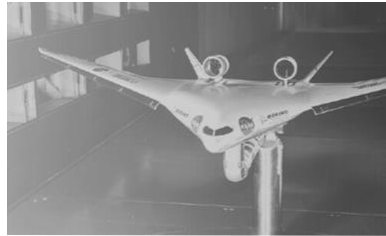




BWB TECHNOLOGY

25 Years of NASA Research

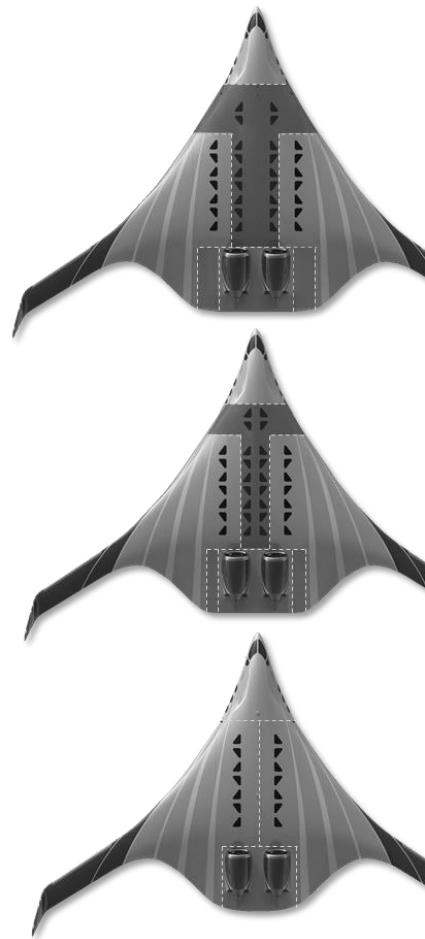
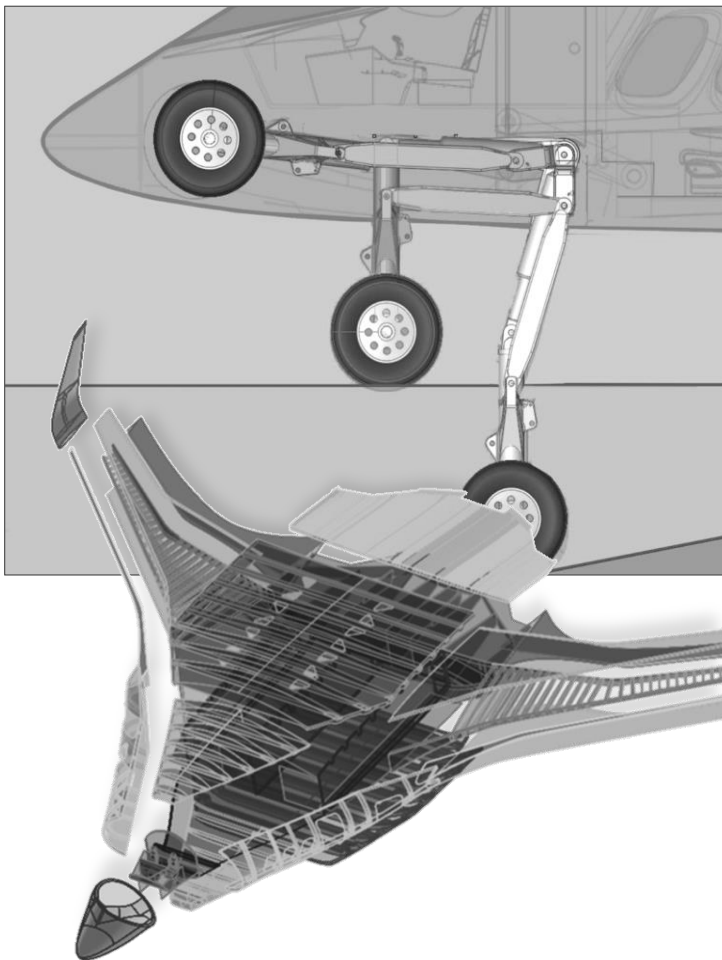
- Transonic Wind Tunnel Tests
- Low-Speed Wind Tunnel Tests
- Powered Wind Tunnel Tests
- Spin and Tumble Tests
- Fuselage Structure Tests
- Flight Tests
- Acoustic Tests
- 15-20% Efficiency Gain
- Achieved TRL 6/7



DZYNE BWB IP

Building since 2012

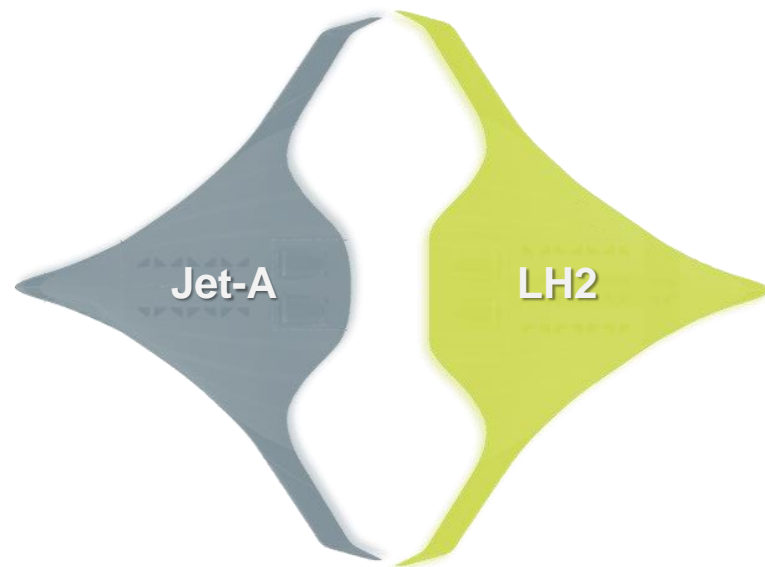
- Single-Deck Design
- Pivot-Piston Landing Gear
- T-Plug Family Growth
- Structural Optimization
- Aerodynamic CFD Optimization
- ANOPP Acoustic Analysis
- NASA/Boeing Tool Correlation
- 30% Efficiency Gain
- New IP = 50% > BWB gain ... TRL 2/3*



KEY ENABLER BWB Enables LH2 Adoption

BWB benefits for LH2 Adoption – Typical Transport

- LH2 Cryo-tanks Located outside the cabin
- Span Small Increase
- TOGW 5% more
- TOFL Slightly longer
- LFL Slightly longer
- Cabin layout No Change
- Loading No change

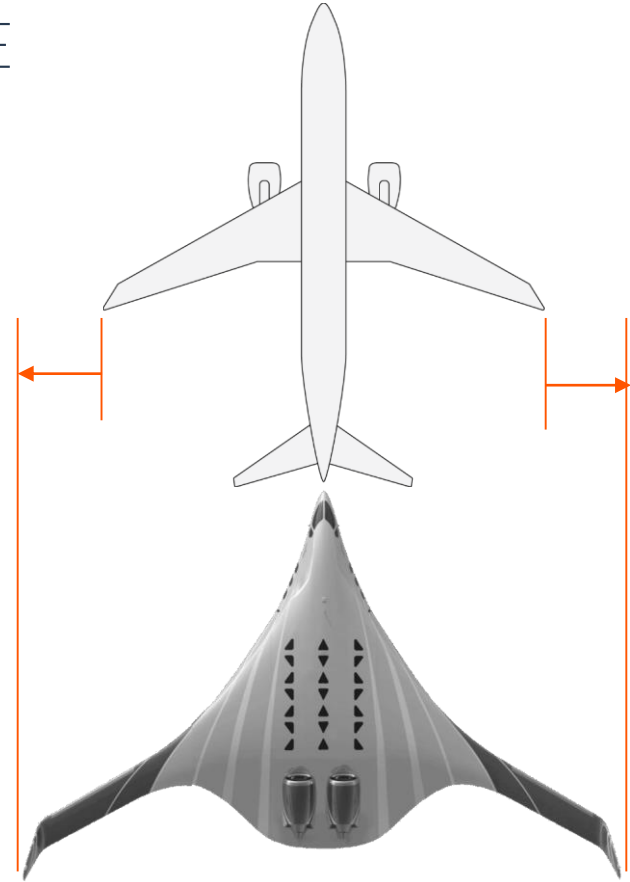


- Far more encouraging than current EASA projections for LH2 adoption

BWB TECHNOLOGY FOR EVERYONE

Significant benefit to all stakeholders

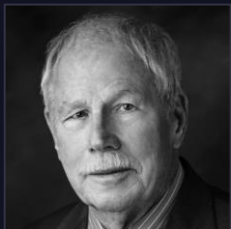
1. Save the Planet with 30% lower emissions
2. Serve the public with dramatically lower noise
3. Serve the Airlines with 30% less fuel-burn



$$\text{Aerodynamic Efficiency} = K * \text{Span} / \sqrt{\text{surface area}}$$

"How do we make BWB a reality"

Technical Advisors



BOB LIEBECK



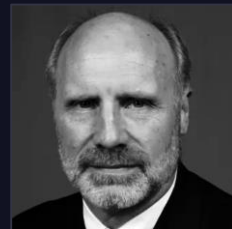
BLAINE RAWDON



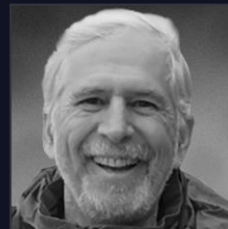
BRYAN MOSS



PRES HENNE

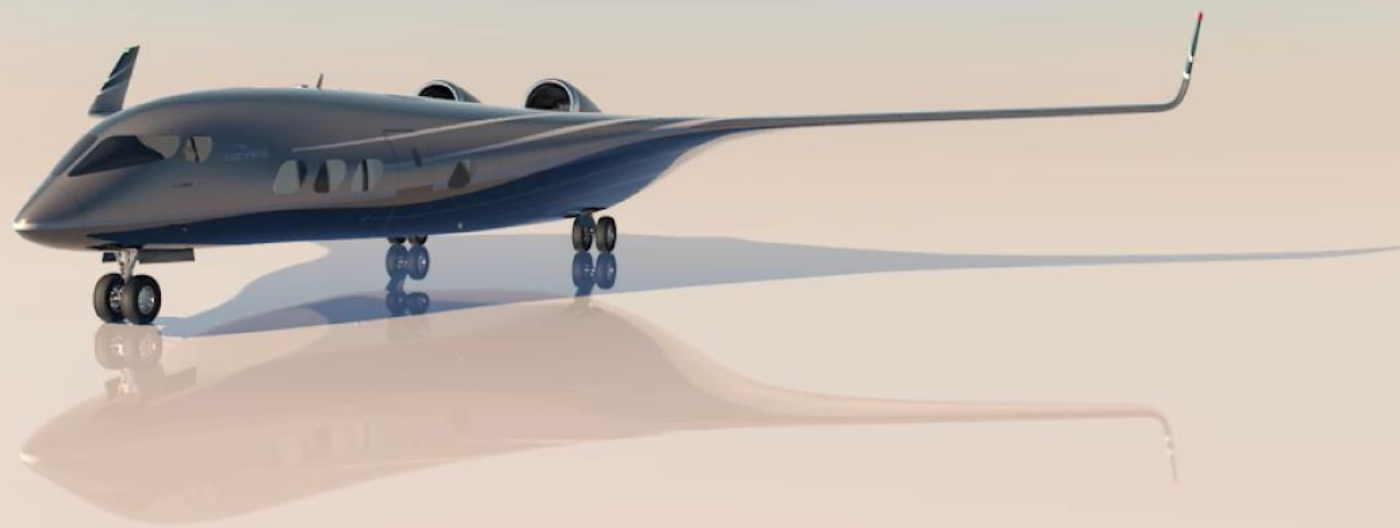


WARREN WILLITS



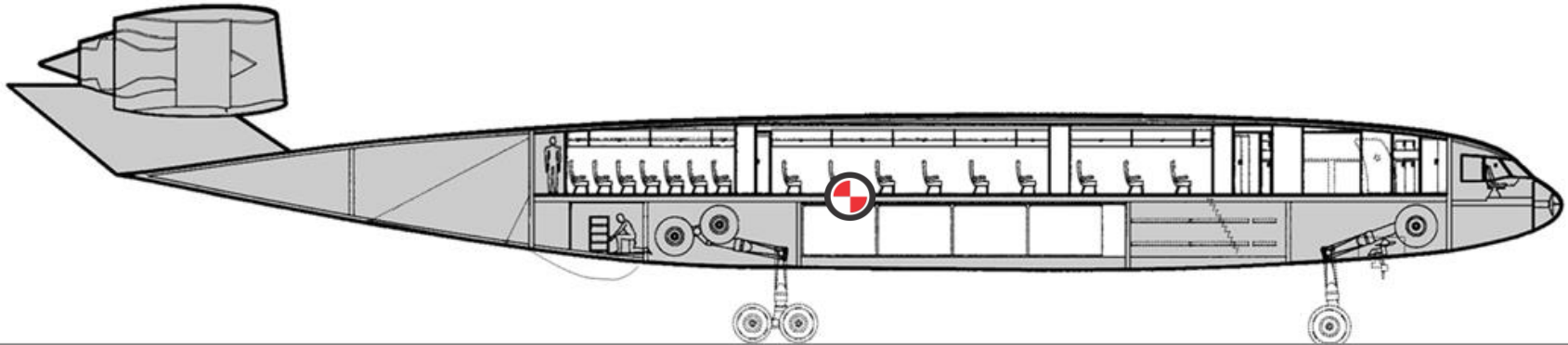
ILAN KROO

“Launch now – finish the research – and Go”



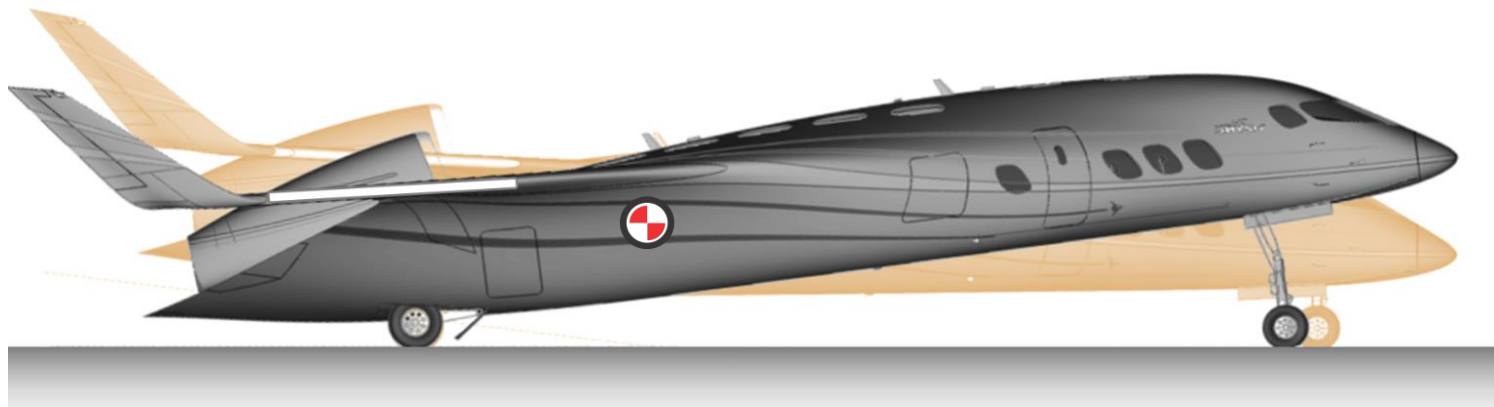
TAKEOFF ROTATION THE FLYING WING CHALLENGE

- Main-gear needs to be near the CG
- This allows rotation and de-rotation
- Solution – main-gear below the cabin
- Double-Deck works, but it makes a big BWB
- Double-Deck too big for “Single-Aisle” market
- Single-Deck needed for smaller BWB’s



PIVOT-PISTON ENABLES A SINGLE-DECK BWB

- Pivot-Piston main-gear is behind cabin
- "Virtual" rotation about the CG
- Nose and Mains hydraulically linked
- Passive hydraulics - no pumps
- Main-gear squat powers nose-gear extension
- Minimal elevon download to rotate the plane



SURPRISE BENEFITS

- Cargo beside cabin = better span-loading
- Reduced wing bending-loads
- Much less elevon needed for rotation
- Better max lift at rotation (CL_{VMU})
- 15kt better liftoff = 2,000ft less ground-roll
- Nosewheel brakes = 30% less stopping distance



PIVOT-PISTON SCALE DEMONSTRATOR

DEMONSTRATION OBJECTIVES

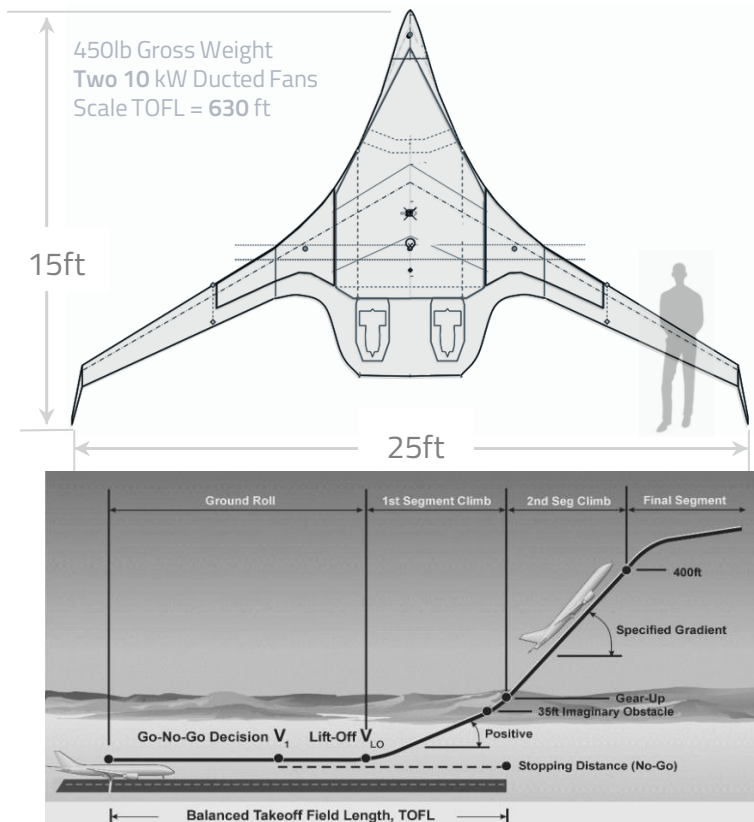
- Pivot-Piston functionality
- Off-design & failure cases
- Performance relative to conventional gear

TEST ARTICLE

- 14% dynamically-scaled flight test model
- Fully articulated Pivot-Piston landing gear

TEST OUTLINE

- Takeoff rotation, lift-off, retraction
- Landing, extension, touchdown, de-rotation
- FAR25 takeoff performance
- Extreme conditions & failures



PIVOT-PISTON FULL-SIZE GEAR DEMO

VALIDATION OBJECTIVES

- Critical FAR25 load conditions
- Shimmy-free
- Fail-safe

TEST ARTICLE

- Full-Scale gear assemblies
- Individual gear test fixtures
- 3 gear armature for combined testing

TEST OUTLINE

- Isolated gear static load, & shimmy
- Total armature rotation & drop testing
- Failure-mode testing

TEST FACILITY

- Example: Goodrich Super Rig, Oakville, Ontario



A380 gear testing at Oakville Super Rig

A NEW GENERATION OF FLIGHT

GEN I - PROPELLER



- Propeller powered
- Low speed, low altitude
- Dirty, very loud, inefficient

GEN II - JET AGE



- Jet powered
- High speed, high altitude
- Dirty, loud, inefficient

GEN III - SUSTAINABLE



- Clean Propulsion
- High speed, high altitude
- Clean, quiet, efficient

"We need your help"

mark@bwai.co



Thank you!